

QuikScat Lesson Learned



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Agenda



- QuikScat Overview
- **■** Science Product
- **■** Partners
- **■** Job Challenges
- **■** Lessons Learned
 - technical
 - people
 - unexpected
 - post launch
- **■** Conclusion







QuikScat Spacecraft

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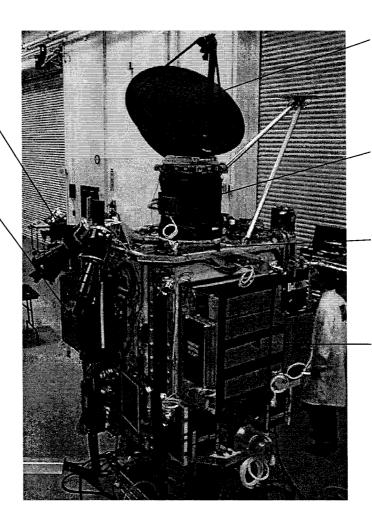


Star tracker

Computing & Data System (CDS)

The SeaWinds on QuikScat mission is a "quick recovery" mission to fill the gap created by the loss of data from the NASA Scatterometer (NSCAT), when the satellite it was flying on lost power in June 1997.





Spinning Antenna

Scatterometer Antenna Subsystem (SAS)

Spacecraft Main Bus

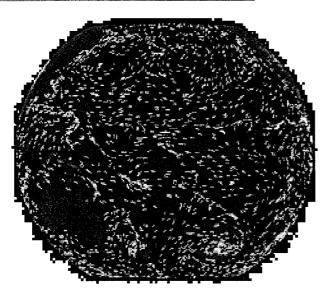
Scatterometer Electronics Subsystem (SES)





Science Product

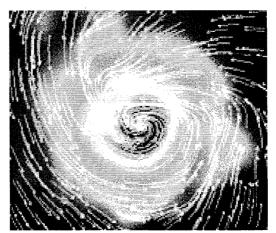




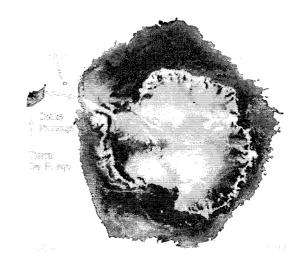
QuikScat acquires all-weather, high-resolution measurements of near-surface winds over global oceans. These measurements improve atmospheric models, weather forecasting, storm warning, ice edge movement, etc...

http://winds/missions/quikscat/quikindex.html





Hurricanes







Partners



- **GSFC programmatic management**
 - First time implementation of NASA's Indefinite
 Delivery/Indefinite Quantity program for rapid delivery of satellite core systems
- BALL Aerospace- spacecraft provider and I&T
- LASP (University of Colorado at Boulder) Mission Operation
- **■** JPL instrument provider and project management
- **■** Other contractors and vendors
 - (Honeywell @ Arizona, E-systems @ Florida and COI @ California) to collaborate for a successful launch from Vandenberg AFB on June 19, 1999







Job Challenges



■ Technical

- launch one year from start
- instrument hardware designed for a different spacecraft - ADEOS II
 - different thermal environments: Beta angles of 58 to 90 vs 12 to 30
 - different physical locations: Nadir deck vs +X,
 +Y & +Z
 - different bus voltage 46 Volts vs 28 Volts







Job Challenges



- **■** Interpersonal
- **■** Immediate team building
 - numerous interchange meetings
 - coherent working environments
 - success oriented approach with little room for errors







Types of Lessons Learned



■ Technical

- interfaces, resources, schedule
- **■** People
 - commitment, acceptance, trust, decision making
- What worked and what didn't work







Technical Lessons Learned



- **■** The dominant constrain was schedule
 - time spend on minimizing budget was a waste of schedule
 - competence saves time, get the "A-Team"
- Seek system wide solutions
 - sub-optimization may cause problems elsewhere
 - holding resources and margin locally makes somebody else's live more difficult
- Work being conscious of the impact that your design has on others
- **■** Work in anticipation of what comes next







Technical Lessons Learned



- Cursory evaluation of min/max beta angles was not adequate.
- Needed to survey worst case seasonal conditions (e.g., SES worst hot at beta=64 deg).
- Needed to identify worst case condition for each subsystem individually (accounting for different orientation).
- Don't mount temperature sensors on heater boards.







People Lessons Learned



- first we built a team, then we built a spacecraft
- **■** we cared for each other
- respect
- **■** trust
- **■** enjoy cooperation
- competence
- **■** reliability
- **■** communication
- person to person interaction (travel)







Unexpected Lessons Learned



Expectation vs Requirements

- exists
- better to clarify before launch instead of 2
 am in the morning after launch
- explain your design and guide expectations
- maintain communication between the design team and the science/project team to ensure common expectations



3/3/2000





Unexpected Lessons Learned



■ About 2 WM worth of post launch analysis was performed. The two issues were temperature and temperature stability.

■ Temperature

- All instrument temperatures were within flight allowables (also within 5C of predictions).
- Worst case predicts were not sufficient. Seasonal behavior was requested. Trend to more "resolution" in the analysis.



3/3/2000



Post Launch Lessons Learned



■ Temperature (cont.)

- Solar panel segments nearest critical SES radiator were turned off causing a warmer thermal environment. Insufficient communication between power and thermal subsystem.
- We are currently flying with solar arrays tilted by30 deg (away from radiator) to reduce SES temperatures to more acceptable levels.



3/3/2000





Post Launch Lessons Learned



■ Stability

- Temperature stability was defined as a design goal of 5C per orbit.
- Worst peak-to-peak variation is 7C, 2C higher than the goal.
- Because of long term reliability concerns much analysis was spend on more detailed understanding.
- Expected a value close to 0C at beta = 90deg but measured 1.2C.
- Need to take variable albedo and planet IR into account.





Post Launch Lessons Learned



■ Applicability of Requirements

- Define where the requirements apply. Be specific about the location and instrument/nodalize accordingly.
- Be explicit about the difference between system level requirements versus subsystem/assembly performance.



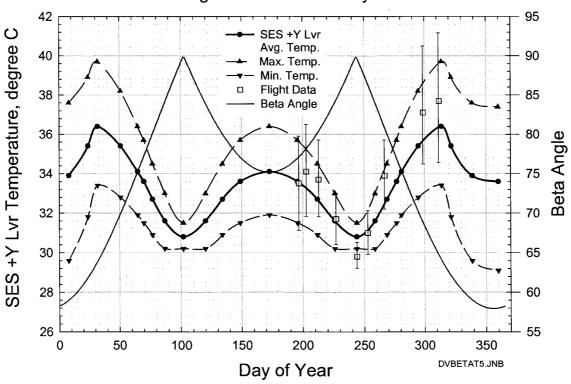




Picture of annual cycle



Figure 9. QuikSCAT Nominal Case SES +Y Lvr Temperatures and Beta Angle Variation with Day of Year in 1999









Conclusions



■ Enormous accomplishments can be made when

- competent people form a real team
- authority to make decisions is commensurate with the level of responsibility
- bureaucracy and budget doesn't constrain engineering
- people are given the opportunity to interact in person (adequate travel budget)
- everybody enjoys his/her work
- YOU HAVE A TEAM LIKE THIS →





